

SUPPLEMENTAL TYPE CERTIFICATE

10014147 REV. 3

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

ATELIER CHABORD

125 ROUTE DE BELLEGARDE
74330 EPAGNY
FRANCE

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.A.367

Type Certificate Holder: CEAPR

Type: DR 200, DR 300, DR 400 SERIES

Model: DR 300/108, DR 315, DR 400/100

DR 400/120, DR 400/120 D

DR 400/2+2

Description of Design Change:

Installation of exhaust system according to Chabord modification STC-DR420-2015-01

Rev. 3: - Enhanced silent blocks

- Alternative anti corrosion surface treatment
- Enhanced material on different parts
- Enhanced shape of central structural part in muffler
- Additional bracket on double rod tube.

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

- The following paragraph(s) at a later amendment: CS 23 Amdt 3 § 607, 611, 783, 807, 867, 1101, 1121, 1123, 1125, 1529.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 24 January 2022



Marco CAPACCIO
Section Manager

Small Aircraft, Balloons & Airships



The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact on the noise database.

Associated Technical Documentation:

Technical Dossier ref. STC-DR420-2015-01 Revision D dated 04 January 2022,
Or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Point 21.A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -